

# CENTRIPETAL FORCE & ENERGY\*

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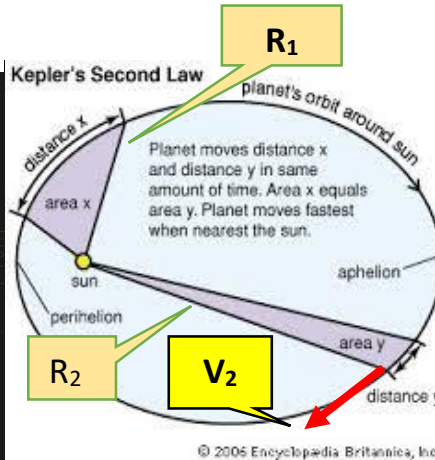
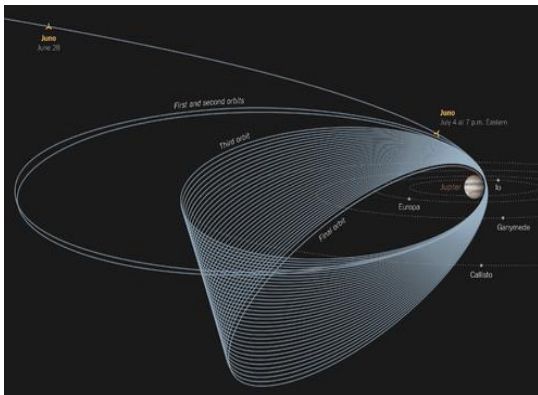
& New York Times , June 28, 2016 by Kenneth Chang

## NASA's Juno Spacecraft Will Soon Be in Jupiter's Grip

**On July 4, 2016, NASA's Juno spacecraft will arrive to study Jupiter after a trip of nearly two billion miles.**

By DENNIS OVERBYE, JONATHAN CORUM and JASON DRAKEFORD on Publish Date June 28, 2016. [Watch in Times Video »](#)

After traveling for five years and nearly 1.8 billion miles, NASA's Juno spacecraft will announce its arrival at Jupiter with the simplest of radio signals: a three-second beep. NASA expects the beep, marking the end of a 35-minute engine burn to slow the spacecraft down and allow it to be captured by Jupiter's gravity, to arrive at Earth at 11:53 p.m. Eastern time next Monday. "I can tell you when that completes, you're going to see a lot of celebration," said Rick Nybakken, Juno's project manager, "because that means we'll be in orbit around Jupiter, and that'll be really cool."



**INTRO:** Kepler's 2<sup>nd</sup>. Law states: planets, moons, satellites sweep Out equal areas in equal amounts of time. Thus in triangles at left:

$$[\frac{1}{2} R_1 x ]/t = [ \frac{1}{2} R_2 y ] / t$$

$$R_1 V_1 = R_2 V_2$$

Jupiter has a 43,441-mile radius and must be considered as part of R. Thus,  $R_2 = 2,043,441$  miles &  $R_1 = 43,441 + 3,100 = 46,541$  miles

**QUESTIONS:** (a) Find  $V_2$  in mph & m/s?

**HINTS:**  $0.44704$  m./s. = 1 mph

**ANSWERS:** (a) 2846.975 mph, 1,272.7 m/s

"Juno is to make a series of 37 highly elliptical orbits passing over Jupiter's north and south poles over 20 months.

**((At its farthest, it will be about two million miles ( $R_2$ ) from Jupiter. For each orbit, it will accelerate inward, reaching 128,000 miles per hour( $V_1$ ), and pass within 3,100 miles( $R_1$ ) of Jupiter's cloud tops)).**

The slight fluctuations in Jupiter's gravitational pull, measured by shifts in the frequency of Juno's radio signals, will tell the density of the planet's interior and whether there is a rocky core within, where pressures might reach half a billion pounds per square inch.



Scott Bolton, the principal investigator for the \$1.1 billion mission, with a model of the solar-powered spacecraft.

**INTRODUCTION 2:** Total energy is conserved of Juno satellite .....

$$K_1 + U_1 = K_2 + U_2, \text{ or } \frac{1}{2} m V_1^2 + [- Gm M_J / R_1 ] = \frac{1}{2} m V_2^2 + [- Gm M_J / R_2 ] \quad \text{eq. 1}$$

Note in eq. 1 instead of  $mgh$  for  $U$ (gravitational potential energy) used is  $-G m M/R$  since this function accounts for changes in gravity with distance.

Our goal here is to show total energy is conserved with data above about:  $V_1, V_2, R_1, R_2$ . Lets rearrange terms in eq. 1 (& remove  $m$  from all terms) to reflect  $\Delta K = \Delta U$  for the fact that total kinetic energy changes should be equivalent to potential energy changes....

$$[ V_1^2 - V_2^2 ]/2 = G M_{\text{Jupiter}} [ 1/R_1 - 1/R_2 ] \quad \text{eq. 2}$$

**QUESTIONS:** (a) Convert Perigee 128,000 mph to m./s. ( $V_1$ )?

(b) Compute left side of eq. 2? (c) Find Perigee  $R_1$  in meters? (d) Find Apogee  $R_2$  in meters? (e) Compute right side of eq. 2? (f) Comment on how well computed energy conservation came out finding kinetic energy change vs. gravitational potential change?

**HINTS:**  $G$  (gravitational constant) =  $6.67408 \times 10^{-11} \text{ N m}^2 / \text{kg}^2$ ,  $M_{\text{Jupiter}} = 1.898 \times 10^{27} \text{ kg}$ .

**ANSWERS:** (a)  $V_2 = 57,221.12 \text{ m./s.}$ , (b)  $1.636 \times 10^9$ , (c)  $R_1$  (perigee) =  $7.49 \times 10^7$  meters (d)  $R_2 = 3.28859 \times 10^9$  meters, (e)  $1.6715 \times 10^9$  (f) Left and right side of eq. 2 are within 1% ( $1.636 \times 10^9$  vs.  $1.6715 \times 10^9$ ). This is quite good considering article data on perigee and apogee are approximations to reality. Convincing energy is conserved.